

A.W. Williams, N.J. Wood, and A.B. Zakaria
Department of Physics, University College of Swansea
University of Wales, Singleton Park, Swansea SA2 8PP U.K.

ABSTRACT

In this paper we use a photothermal imaging technique to characterize the damage caused to an imperfectly coated gold-coated Kapton sample exposed to successively increased fluences of atomic oxygen in a laboratory atomic source.

INTRODUCTION

One major problem associated with the flight of low Earth orbit (LEO) spacecraft is the damage caused to various materials by bombardment with atomic oxygen (AO). AO will readily oxidize materials with high erosion yield coefficients, such as polyimide Kapton, epoxy graphite, and Mylar. Materials with low erosion yield coefficients such as aluminum, gold, and SiO_2 may be used as barrier coatings to prevent damage to the more vulnerable underlying materials mentioned above (ref. 1). However, manufacturing defects in the barrier coatings such as scratches and pin holes act as sites where the AO can attack the substrate, and may cause undercutting of the protective layer (ref. 2).

Photothermal imaging of solids is a powerful technique which has been applied to numerous problems involving the characterization of surface and subsurface, cracks inclusions, and delamination in materials (ref. 3, 4). We have used this method to produce photothermal images of a gold-coated Kapton sample at various stages of its exposure to AO in a laboratory AO source. The gold layer was deliberately scratched prior to exposure of the sample, and the resultant damage around this site imaged photothermally.

PHOTOTHERMAL IMAGING APPARATUS

A diagram of the apparatus is given in Figure 1.

In order to produce a photothermal image, a modulated and localized heat source is required which is used to create thermal waves in the sample. In our system, we have used an argon ion laser as a heat source whose beam is focused by means of a microscope objective onto the sample to be scanned. The laser was operated on a wavelength of 488 nm in order to optimize the absorption in the gold (≈ 63 percent at this wavelength) and thus generates "thermal waves" within the sample being illuminated.

The depth to which the thermal waves penetrate the sample μ_s is given by:

$$\mu_s = \left(\frac{\alpha_s}{\pi f} \right)^{1/2},$$

where α_s is the thermal diffusivity of the material to be investigated, and f is the modulation frequency of the laser light.

The value of μ_s and, hence, the penetration depth may be changed by altering the modulation frequency.

Associated with the thermal waves are acoustic waves which are stresses generated in the sample due to the thermal waves and which penetrate the whole sample and are detected by a piezoelectric transducer coupled to the rear side of the sample. The voltage produced by the transducer, which depends on the magnitude of these waves, is then amplified and detected by a lock-in amplifier (Stanford SR530). The lock-in amplifier gives a reading of the magnitude of the photothermal signal detected, together with its phase shift with respect to the light modulation. The sample together with its piezodetector was mounted upon two orthogonal translation stages and raster scanned beneath the focused laser beam of diameter $\approx 2.5 \mu\text{m}$ in steps of $3 \mu\text{m}$. The power density of the focused laser beam was kept below the damage threshold of the sample.

The thermal waves traveling into the bulk of the sample are reflected and scattered by regions of differing thermal properties within the sample. The photothermal signal, therefore, depends upon these imperfections, and hence gives the imaging capability of this technique.

In order to produce a subsurface image, the X-Y scans consisting of 75 by 75 data points for both the signal and its phase lag were recorded across a small area of the sample. The photothermal signal is sensitive to surface optical features which have differing optical absorption. However, the phase lag is much less sensitive to surface features and is a better measure of subsurface features especially delaminations (ref. 5). We have, therefore, concentrated upon the phase measurements in the results given in this paper.

SAMPLE EXPOSURE

A 1- by 3-cm, 130- μm thick Kapton sample was vacuum coated with 40 nm of gold and an area of 1 by 1 cm selected. A strip of the gold about 35- μm wide and extending from one side of the sample to the other was removed with a thin metal probe to expose the Kapton substrate beneath it. The sample was then mounted in a laboratory AO apparatus similar to the design described by Neely (ref. 6) and exposed to an AO flux of $\approx 1.5 \times 10^{17}$ atoms/cm²/s at a temperature of 200 °C.

The sample was exposed to AO for four successive exposure times, with cumulative fluences of:

- (i) 8.3×10^{20} atoms/cm²
- (ii) 17×10^{20} atoms/cm²
- (iii) 25×10^{20} atoms/cm²
- (iv) 50×10^{20} atoms/cm².

The sample was scanned photothermally before and after each exposure.

RESULTS

Five photothermal scans of the area containing the exposed Kapton, each comprising 75 by 75 data points were obtained giving a photothermal image 220 by 220 μm in area. The photothermal signal was detected by a piezotransducer on the Kapton surface remote from the gold. An imperfection in the gold/Kapton interface, such as delamination, which would introduce an air layer between the gold and Kapton substrate will scatter the thermal waves before they are detected, modifying the photothermal phase, since this is in effect the delay in generation of the photothermal signal with respect to the modulation frequency, hence making the scan sensitive to subsurface imperfections. The data were represented in false colors in the images of the scans, using Unimap 2000, Uniras A/S (ref. 7).

A modulation frequency of 3 kHz was chosen for all the scans which correspond to a thermal wave probe depth of $\approx 3.2 \mu\text{m}$ into the Kapton. Photothermal images of the phase lag of the signal are shown in Figure 2. These images show undercutting and delamination around the bare Kapton section of the sample. A greater phase lag is seen in the scans for the areas in which the gold is no longer in contact with the Kapton. As the sample becomes progressively more damaged, the area around the scratch area is seen to widen and the associated phase lag becomes greater.

Further photothermal analysis, performed in Swansea, will concentrate on coated samples that are scanned while inside the AO apparatus so that the progressive damage may be characterized in situ and in real time.

ACKNOWLEDGMENTS

The work was supported by the Strategic Defense Initiation Organization Office of Innovative Science and Technology (SDIO/IST) under contract number 60921-86-CA226 with the Naval Surface Warfare Center. ABZ wishes to thank the Public Service Department of Malaysia and the Universiti Pertanian Malaysia for the award of a postgraduate scholarship.

REFERENCES

1. Banks, B.A., Rutledge, S.K., Paulsen, P.E., and Streuber, T.J.: NASA Technical Memorandum 101971, 1989.
2. Pippin, H.G.: *Surface Coatings Technology*, vol. 39, No. 40, 1989, p. 595.
3. Wong, Y.H.: in *Scanned Image Microscopy*, Academic Press, London, 1980.
4. Sawada, T., and Kasai, M.: in *Photoacoustic and Thermal Wave Phenomena in Semiconductors*, vol. 3, A. Mandelis (ed.), North Holland, New York, 1987.
5. Zhang, S., and Chen, L.: in *Photoacoustic and Photothermal Phenomena in Semiconductors*, vol. 27, A. Mandelis (ed.), North Holland, New York, 1987.
6. Neely, W.C., Yang, T.C., Wey, J.P., Clothiaux, E.J., and Worley, S.D.: "Space Structures, Power, and Power Conditioning," SPIE, vol. 871, 1988, p. 313.
7. Unimap 2000 User Manual, Uniras A/S, Søborg, Denmark.

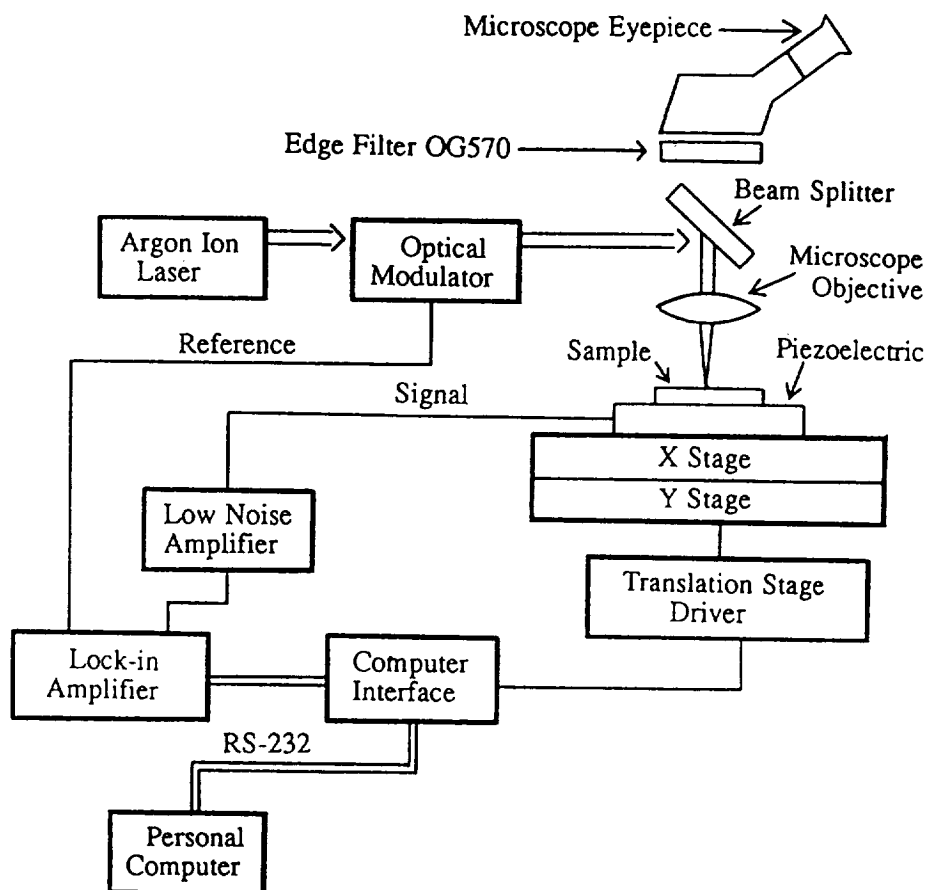


Figure 1. Diagram of photothermal imaging apparatus.

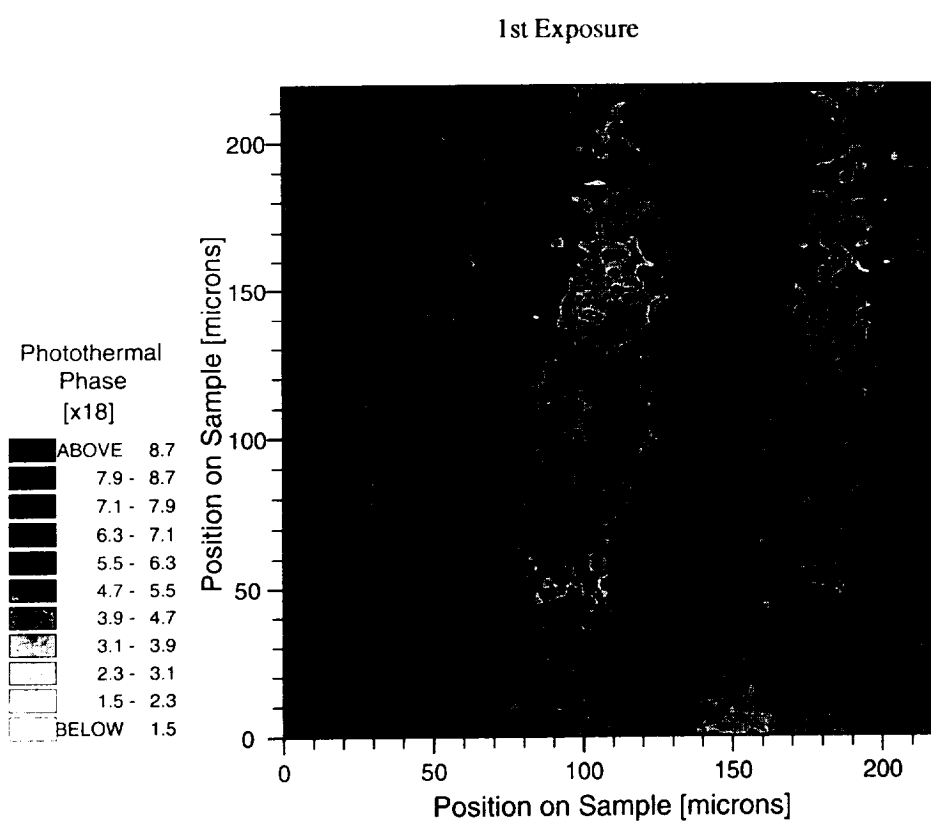
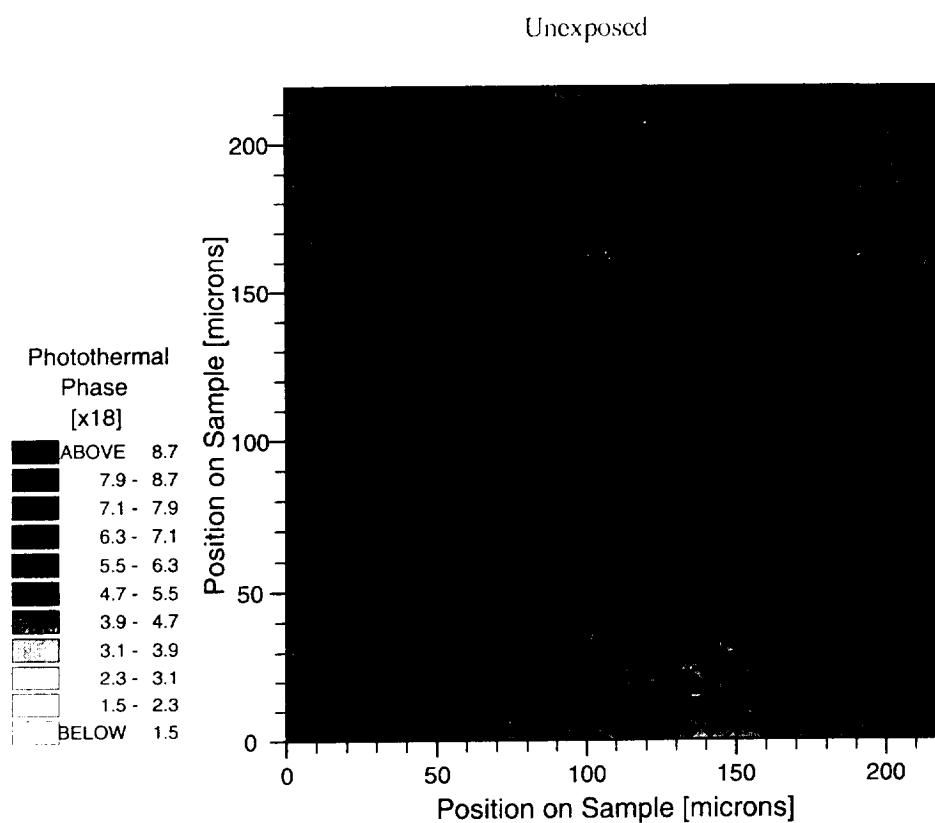
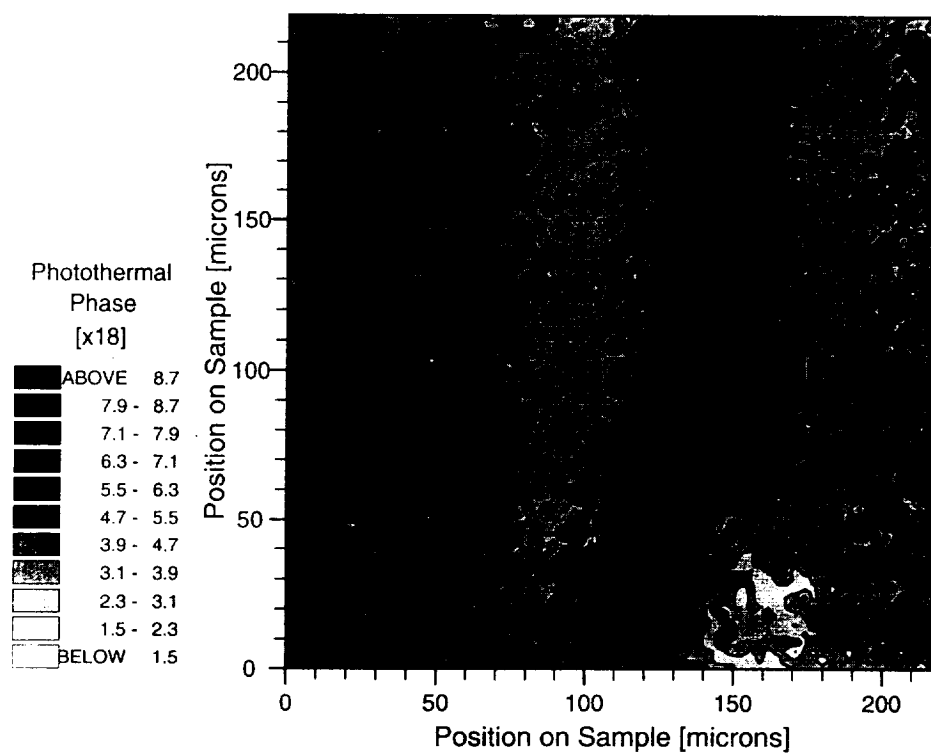


Figure 2. Photothermal phase images for cumulative AO exposures of gold-coated Kapton.

2nd Exposure



3rd Exposure

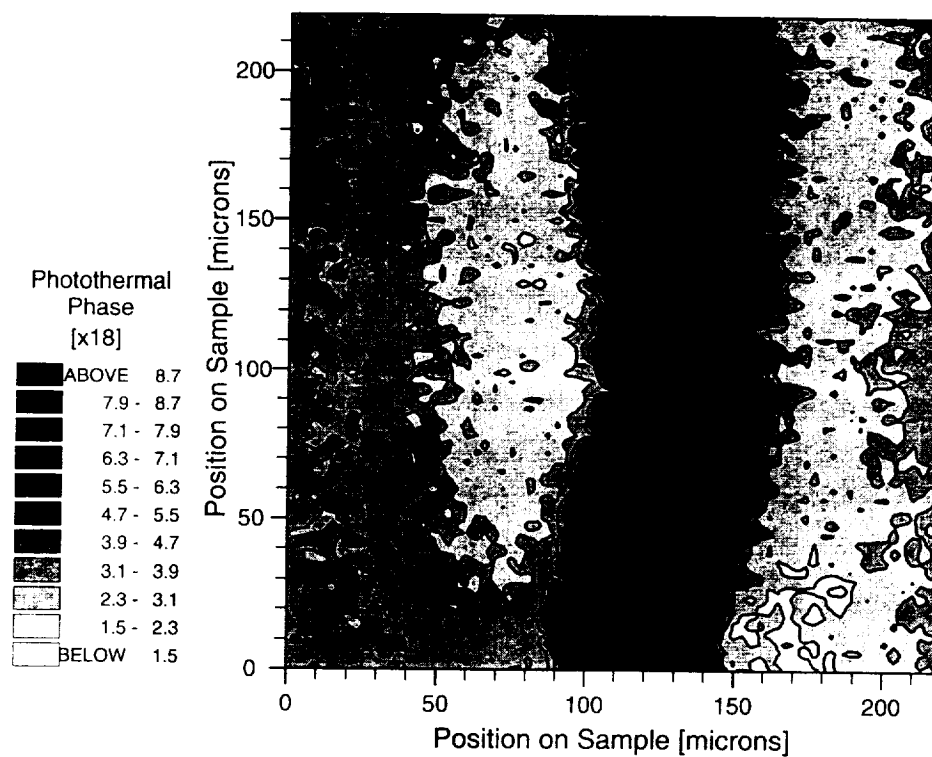


Figure 2. Photothermal phase images for cumulative AO exposures of gold-coated Kapton (continued).

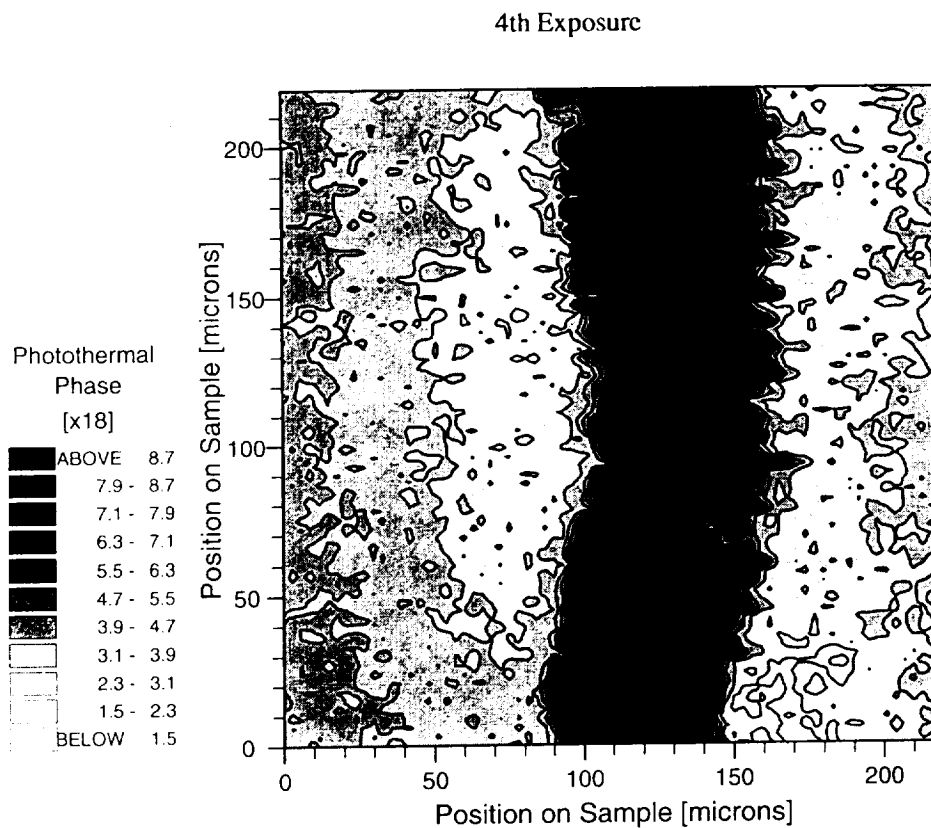


Figure 2. Photothermal phase images for cumulative AO exposures of gold-coated Kapton (continued).

REPORT DOCUMENTATION PAGEForm Approved
OMB No. 0704-0188

Public reporting burden for this collection of information is estimated to average 1 hour per response, including the time for reviewing instructions, searching existing data sources, gathering and maintaining the data needed, and completing and reviewing the collection of information. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing this burden, to Washington Headquarters Services, Directorate for Information Operations and Reports, 1215 Jefferson Davis Highway, Suite 1204, Arlington, VA 22202-4302, and to the Office of Management and Budget, Paperwork Reduction Project (0704-0188), Washington, DC 20503.

1. AGENCY USE ONLY (Leave blank)**2. REPORT DATE**

December 1993

3. REPORT TYPE AND DATES COVERED

Conference Publication

4. TITLE AND SUBTITLE

LDEF Materials Results for Spacecraft Applications

5. FUNDING NUMBERS**6. AUTHOR(S)**

Ann Whitaker and John Gregory, Compilers

7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES)George C. Marshall Space Flight Center
Marshall Space Flight Center, AL 35812**8. PERFORMING ORGANIZATION
REPORT NUMBER**

M-742

9. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES)National Aeronautics and Space Administration
Washington, DC 20546-0001**10. SPONSORING/MONITORING
AGENCY REPORT NUMBER**

NASA CP-3257

11. SUPPLEMENTARY NOTES Proceedings of a conference sponsored by the National Aeronautics and Space Administration, Washington, DC, The University of Alabama in Huntsville, The Alabama Space Grant Consortium, and the Society of Advanced Materials and Process Engineers, and held in Huntsville, Alabama, October 27-28, 1992.

12a. DISTRIBUTION/AVAILABILITY STATEMENTUnclassified - Unlimited
Subject Category: 12**12b. DISTRIBUTION CODE****13. ABSTRACT (Maximum 200 words)**

These proceedings describe the application of LDEF data to spacecraft and payload design, and emphasize where space environmental effects on materials research and development is needed as defined by LDEF data. The LDEF six years of exposure of materials has proven to be by far the most comprehensive source of information ever obtained on the long-term performance of materials in the space environment. The conference provided a forum for materials scientists and engineers to review and critically assess the LDEF results from the standpoint of their relevance, significance, and impact on spacecraft design practice. The impact of the LDEF findings on materials selection and qualification, and the needs and plans for further study, were addressed from the NASA, DoD, industry, and academic perspectives. Many timely and needed changes and modifications in external spacecraft materials selections have occurred as a result of LDEF investigations.

The Executive Summary of this conference is being printed as CP-3261.

14. SUBJECT TERMS

Materials, Environments, Spacecraft, Space Environment Effects

15. NUMBER OF PAGES

562

16. PRICE CODE

A24

**17. SECURITY CLASSIFICATION
OF REPORT**

Unclassified

**18. SECURITY CLASSIFICATION
OF THIS PAGE**

Unclassified

**19. SECURITY CLASSIFICATION
OF ABSTRACT**

Unclassified

20. LIMITATION OF ABSTRACT

Unlimited